



Thermal Control Subsystem FAME

PRELIMINARY TEMPERATURE PREDICTIONS

- ASSUMPTIONS
- CASES
- RESULTS
- CONCLUSIONS
- FORWARD WORK



ASSUMPTIONS



- ALTITUDE = 19323.4 NM
- ENVIRONMENTAL CONSTANTS:
SOLAR = 444 W/in² ALBEDO = .28% IR = 70 W/in²
- BLANKET α/ϵ = .37/.78
- SCT α/ϵ = .10/.85
- TRIM TABS:
 - TILTED 45° FROM SPIN AXIS.
 - SCT ON BOTH SIDES - NO BLANKETS.
- SHEET METAL SUN SHADE WITH SCT ON EXPOSED SIDES - DEPENDING ON CASE.
- SOLAR PANELS:
 - 24.72% SOLAR CELL COVERAGE IN NO PITCH CASE.
 - 34.9% SOLAR CELL COVERAGE IN 10° PITCH CASE.
 - UNPOPULATED/UNBLANKETED AREA IS SCT.



CASES



- PANELS NORMAL TO SPIN AXIS
- PANELS PITCHED 10° TOWARD INSTRUMENT
- PANELS AND SUNSHADE BLANKETED
- PANELS AND SUNSHADE UNBLANKETED



RESULTS

(1 of 6)

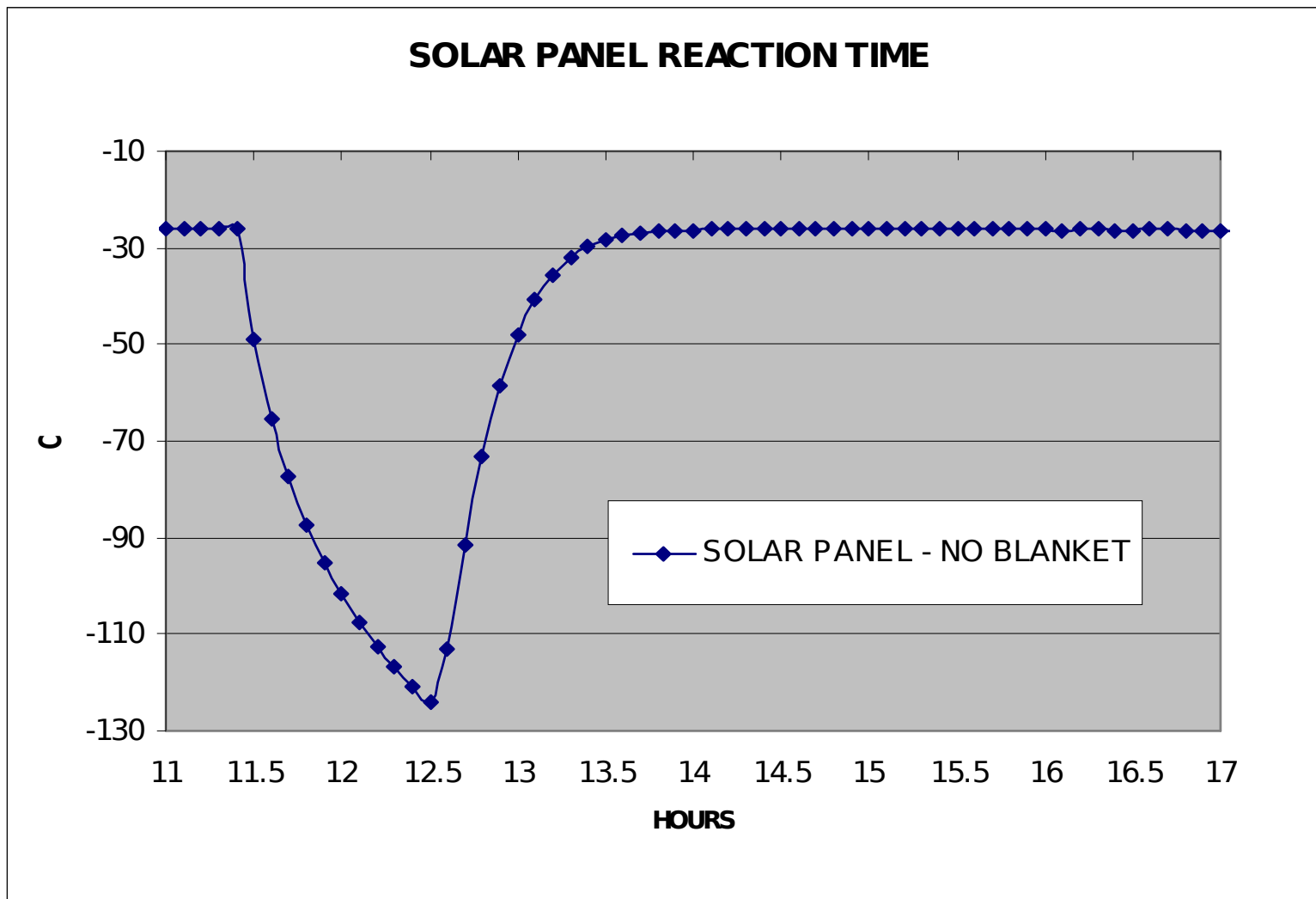


	S/C	S/A		Sun Shield		
<u>CASE STUDY</u>	BUS MLI	Sun Side	Shade Side	Sun Side	Shade Side	Trim Tabs
CASE 1 - NO PITCH w/ MLI						
Solar Cells Cover 24.72% of S/A	-165	-5	-142	-70	-162	n/a
Orbital Min/Max	n/a	n/a	n/a	n/a	n/a	-150 / -78
CASE 2 - NO PITCH w/o MLI						
Solar Cells Cover 24.72% of S/A	-101	-40	-41	-91	-92	n/a
Orbital Min/Max	n/a	n/a	n/a	n/a	n/a	-115 / -85
CASE 3 - 10° PITCH w/ MLI						
Solar Cells Cover 34.9% of S/A	-163	18	-132	-65	-170	n/a
Orbital Min/Max	n/a	19 / 17	n/a	-68 / -62	n/a	-150 / -78
CASE 4 - 10° PITCH w/o MLI						
Solar Cells Cover 34.9% of S/A	-90	-25	-26	-91	-92	n/a
Orbital Min/Max	n/a	-28 / -22	-29 / -23	-97 / -85	-98 / -86	-112 / -83



Results

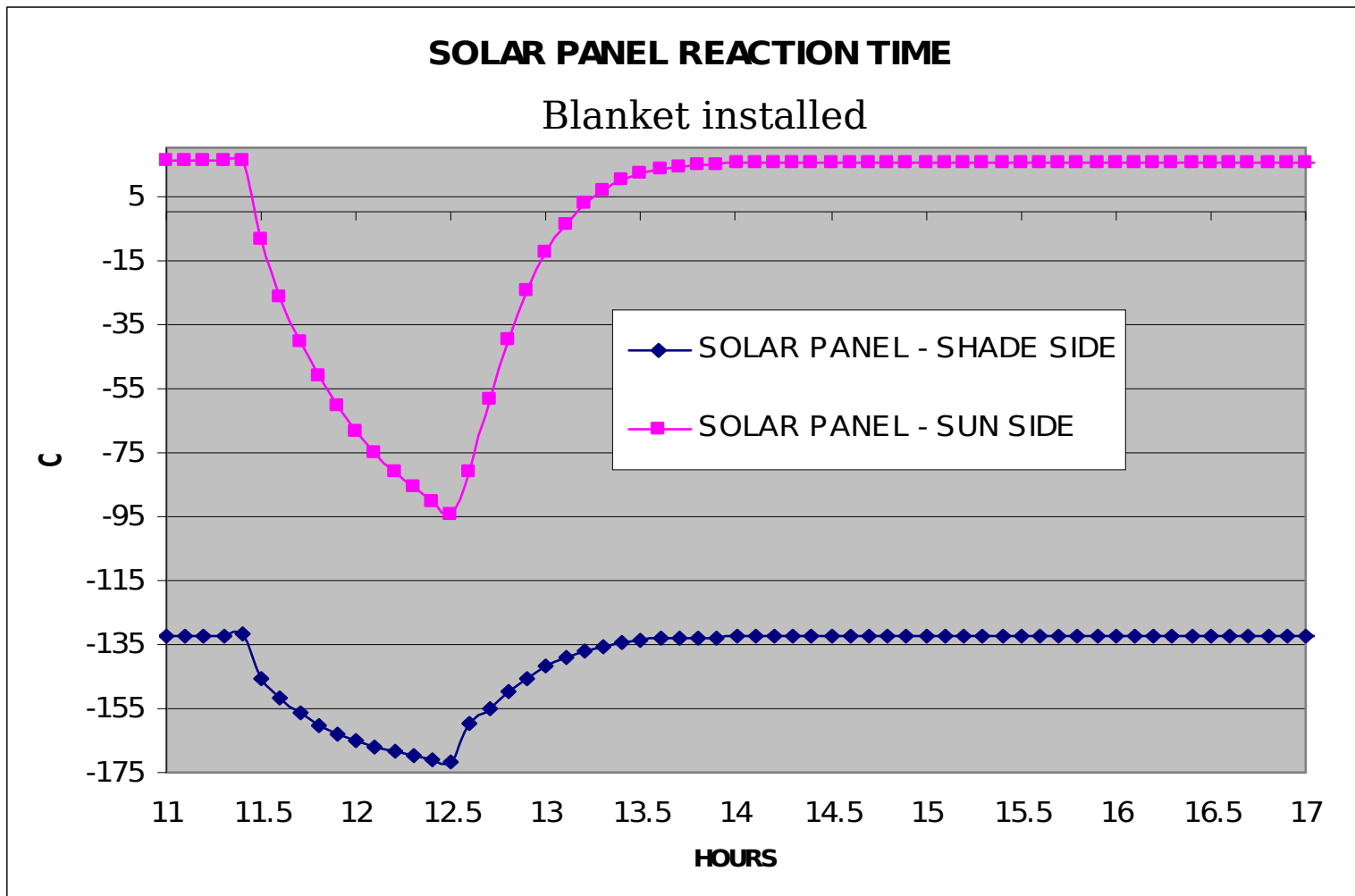
(2 of 6)





Results

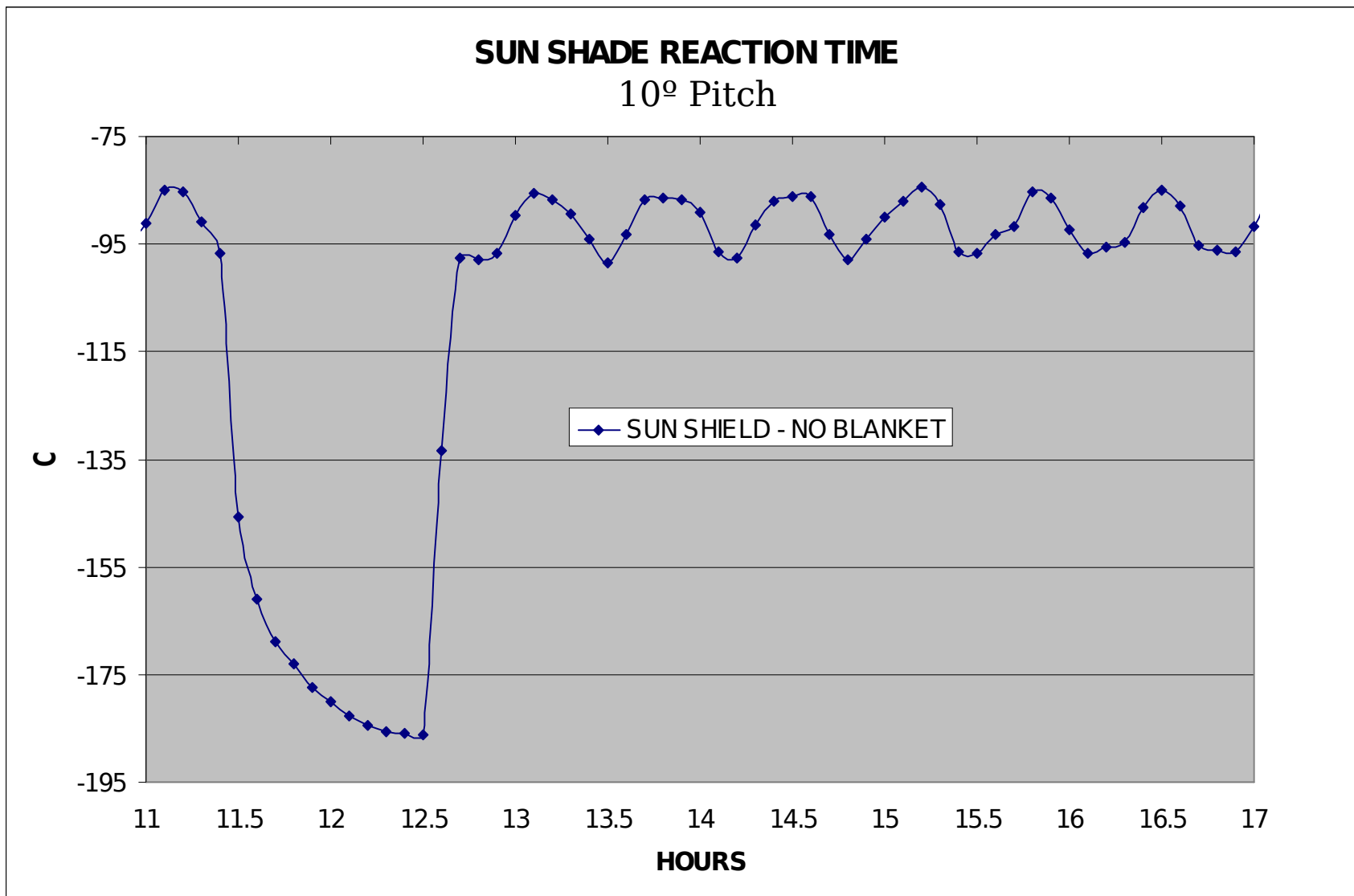
(3 of 6)





Results

(4 of 6)



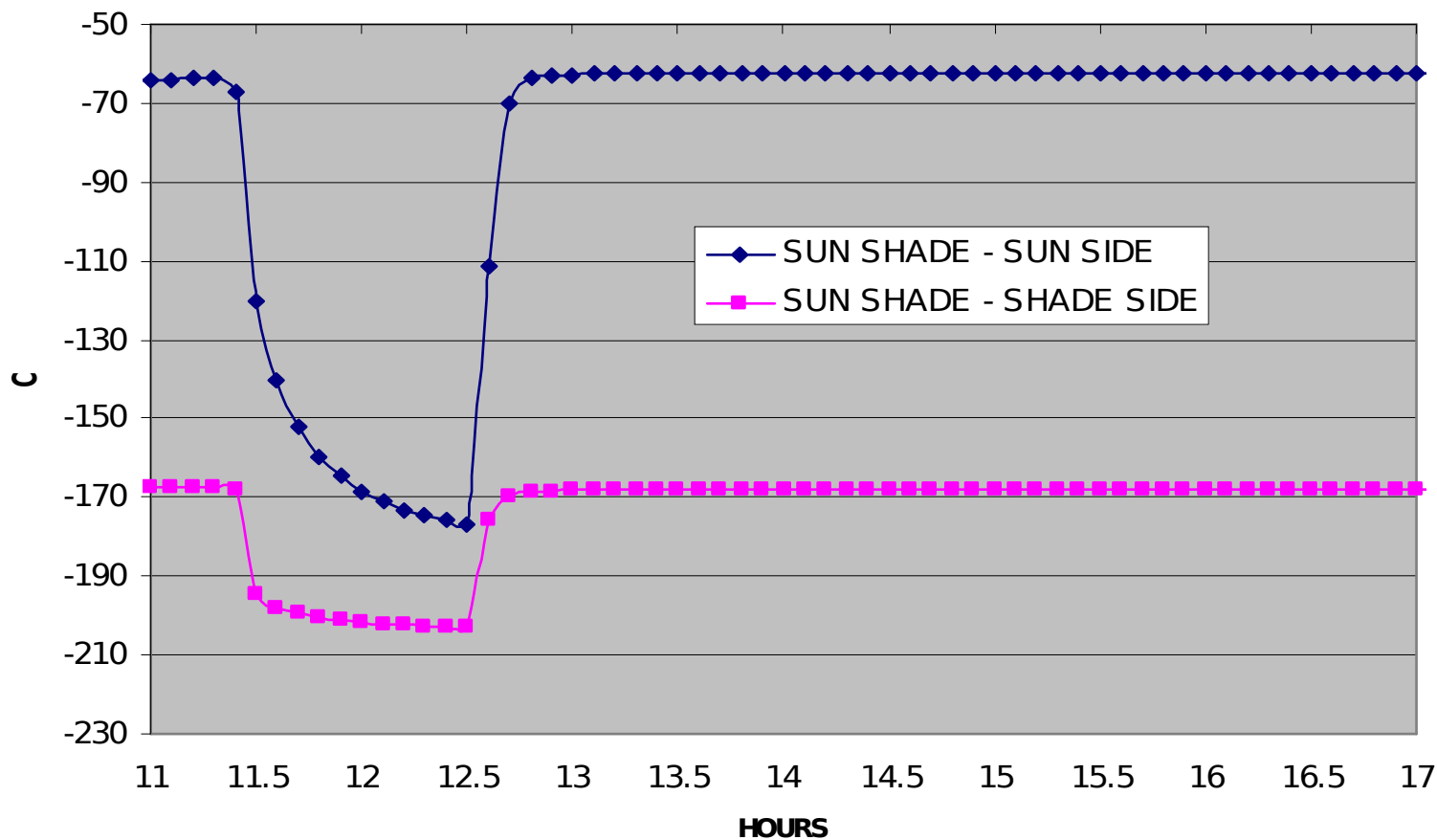


Results

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SUN SHADE REACTION TIME
Blanket installed / 10° Pitch



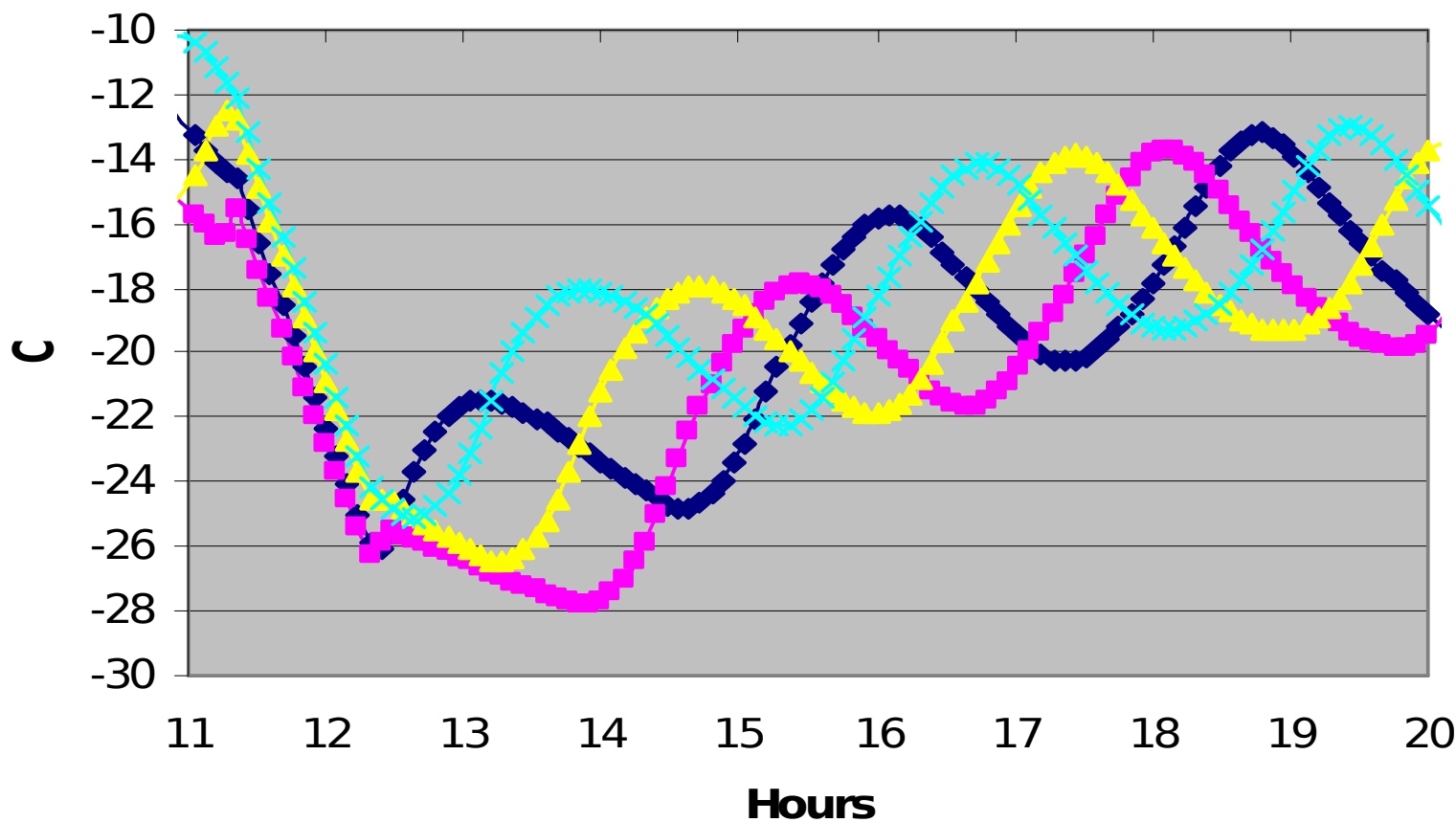


Results

(6 of 6)



AKM Cavity Reaction Time





CONCLUSIONS



- Trim Tab temperature swing less severe when panels are blanketed.
 - May have biggest impact on instrument since temperature swing will be apparent over entire mission – view to instrument.
- About 3 hours for the vehicle temperature to return to pre-eclipse state (passively).
 - Vehicle stability an affected yet separate issue.
- Blankets on sun shield and panels cause more severe temperature changes during eclipse for those components.
 - NO MLI / with MLI
 - 100 vs 113°C ΔT for solar panels.
 - 90 vs 113°C ΔT for sun shield.
- I have a lot of work to do.....(see next slide).



FORWARD WORK

(1 of 2)



- Begin running worst hot/cold cases.
 - Worst case environmental constants, blanket emissivities, BOL/EOL material properties, min/max line voltages.
- Size electronics deck radiator.
 - This will determine required Heater circuit dissipations /number of circuits.
 - Thermal time constant – Reaction time to regain stability.
 - Box layout on deck.
- Add detail to Instrument.
 - In order to attain Interface Heater/Conductance requirement.
 - Predict star tracker interface/heater requirement.
 - Get fluxes on Instrument apertures.
 - Antenna temperature prediction for required test limits.



FORWARD WORK

(2 of 2)



- Incorporate realistic solar cell layout.
- Geometry changes
 - Trim Tab/Bus size.
 - Box layout
- Verify all conductors/masses